

Global astrometry with GAIA Rudolf Le Poole

Abstract

- Why
 - Scientific case:
 - Milky Way : Structure, Dynamics, Star Formation
 - Stellar Physics : Luminosities, Ages, Masses
 - Solar System
 - Extragalactic: Galaxies, Quasars, Reference Frame
 - Fundamental Physics
- How
 - Mission Concept, and Design Considerations
 - Global vs Narrow Field Astrometry
 - Basic Angle
 - Scanning Strategy
 - Limiting Magnitudes
 - Limiting Accuracies
 - Completeness
 - Radial Velocities
 - Photometry
 - Payload
 - Configuration
 - Optical Layout
 - Focal Planes
 - On Board Object Detection
 - Data Analysis
 - Requirements and concept
- Mission and Spacecraft requirements
 - Mission Duration
 - Orbit
 - Thermal
 - Mass
 - Power
 - Telemetry



GAIA

Composition, Formation and Evolution of our Galaxy

Cornerstone Presentations, Paris, 13 September 2000

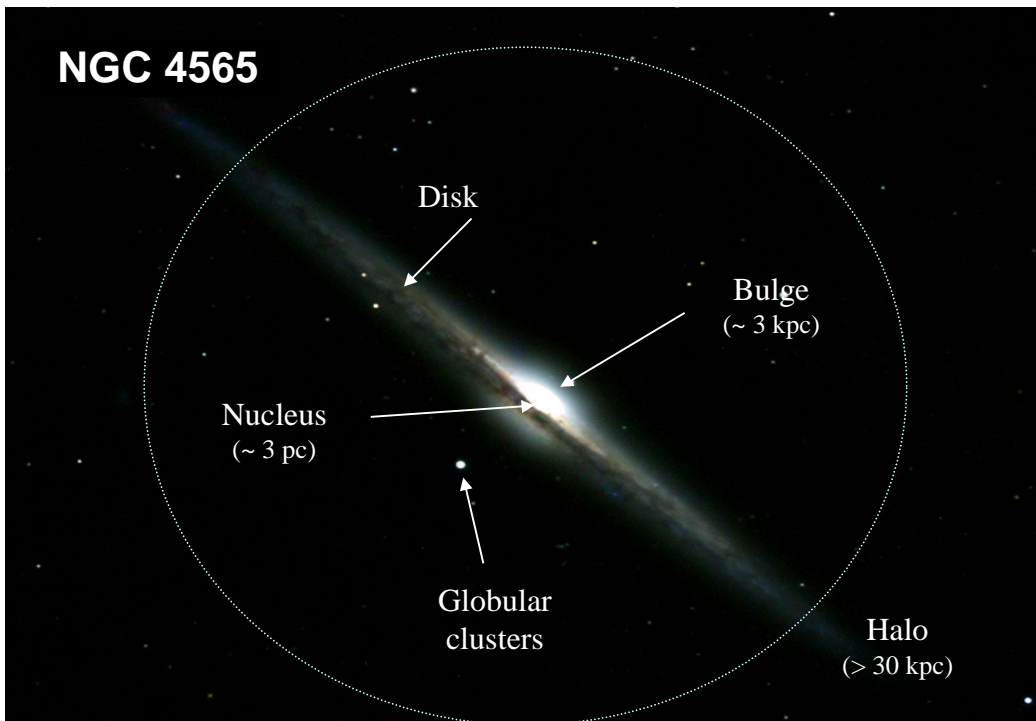
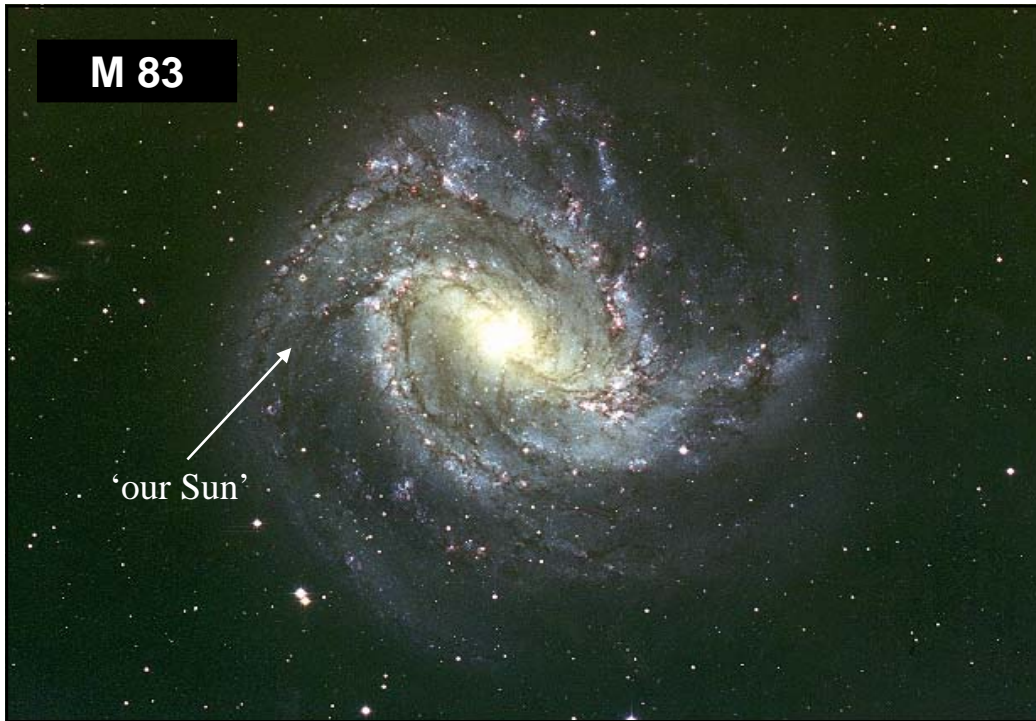
Scientific Case:	P. T. de Zeeuw
Payload, Accuracy and Data Analysis:	L. Lindegren
Spacecraft and Mission Implementation:	O. Pace
Why, How and When:	G. Gilmore

...on behalf of the GAIA Science Advisory Group, GAIA working groups,
ESA Future Projects & Astrophysics Division, and industrial teams

GAIA Science Advisory Structure

- Science Advisory Group members (1997-2000):

K.S. de Boer (Bonn)	L. Lindegren (Lund)
G. Gilmore (Cambridge)	X. Luri (Barcelona)
E. Hoeg (Copenhagen)	F. Mignard (Grasse)
M.G. Lattanzi (Torino)	P.T. de Zeeuw (Leiden)
- Associated Working Groups:
 - Science Working Group (17 members)
 - Instrument Working Group (16 members)
 - Photometry Working Group (17 members)
 - Members at Large (47 members)
- ESA:
 - O. Pace (Future Projects, Study Manager)
 - M.A.C. Perryman (Astrophysics Division, Study Scientist)



GAIA: Key Science Objectives

- Structure and kinematics of our Galaxy:
 - shape and rotation of bulge, disk and halo
 - internal motions of star forming regions, clusters, etc
 - nature of spiral arms and the stellar warp
 - space motions of all Galactic satellite systems
- Stellar populations:
 - physical characteristics of all Galactic components
 - initial mass function, binaries, chemical evolution
 - star formation histories
- Tests of galaxy formation:
 - dynamical determination of dark matter distribution
 - reconstruction of merger and accretion history

⇒ Origin, Formation and Evolution of the Galaxy

GAIA: Paris, 13 September 2000

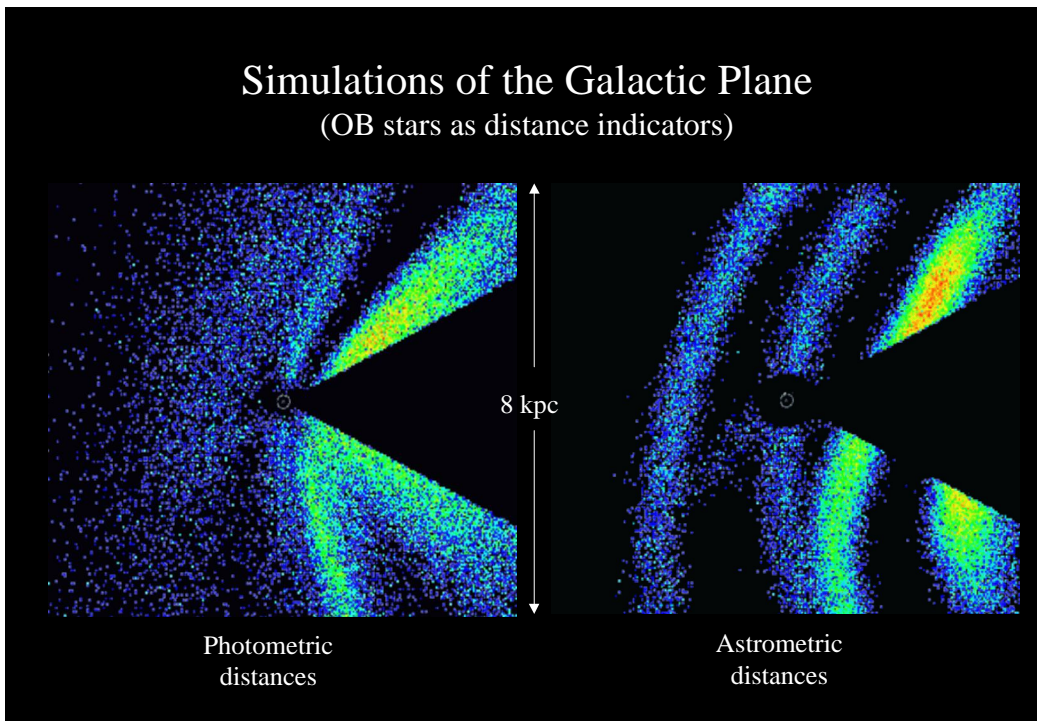
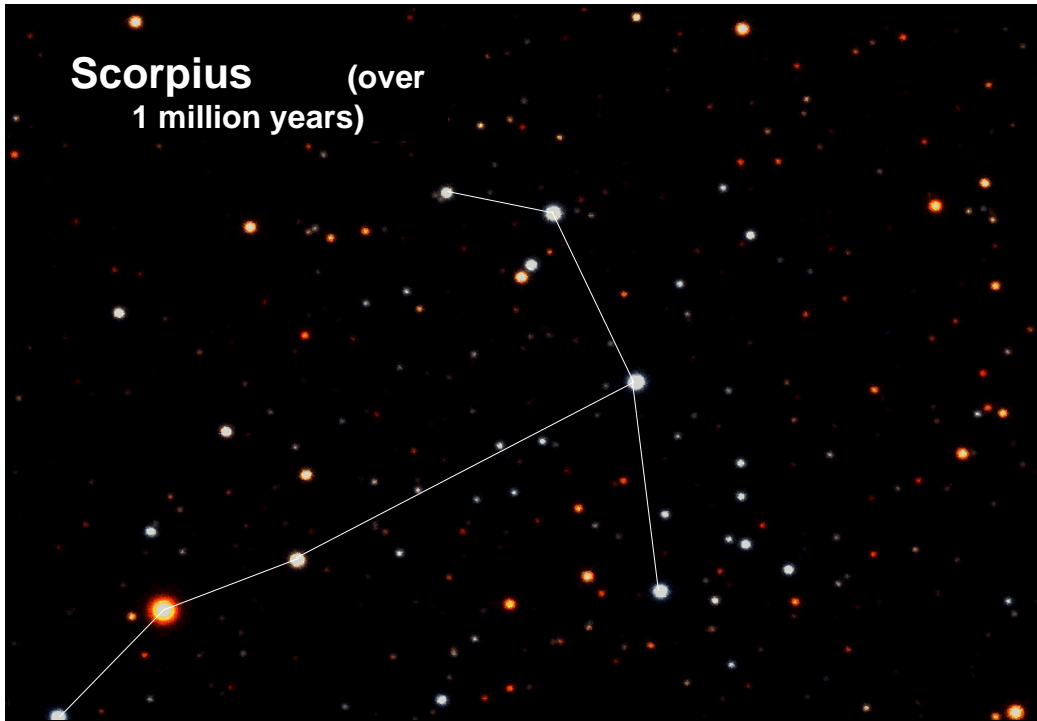
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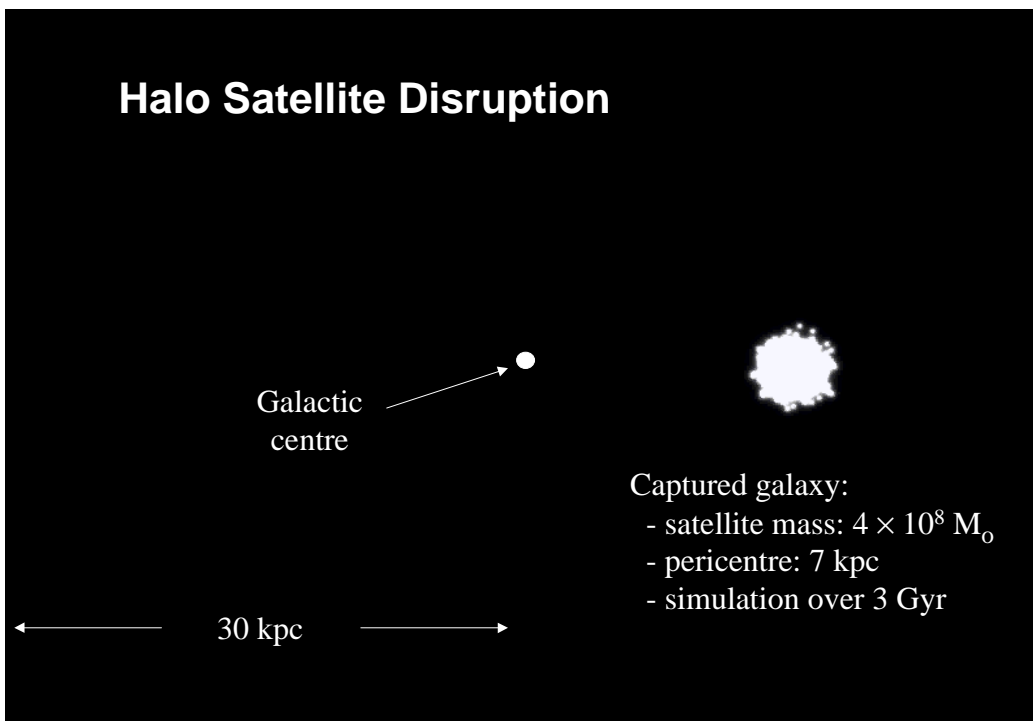
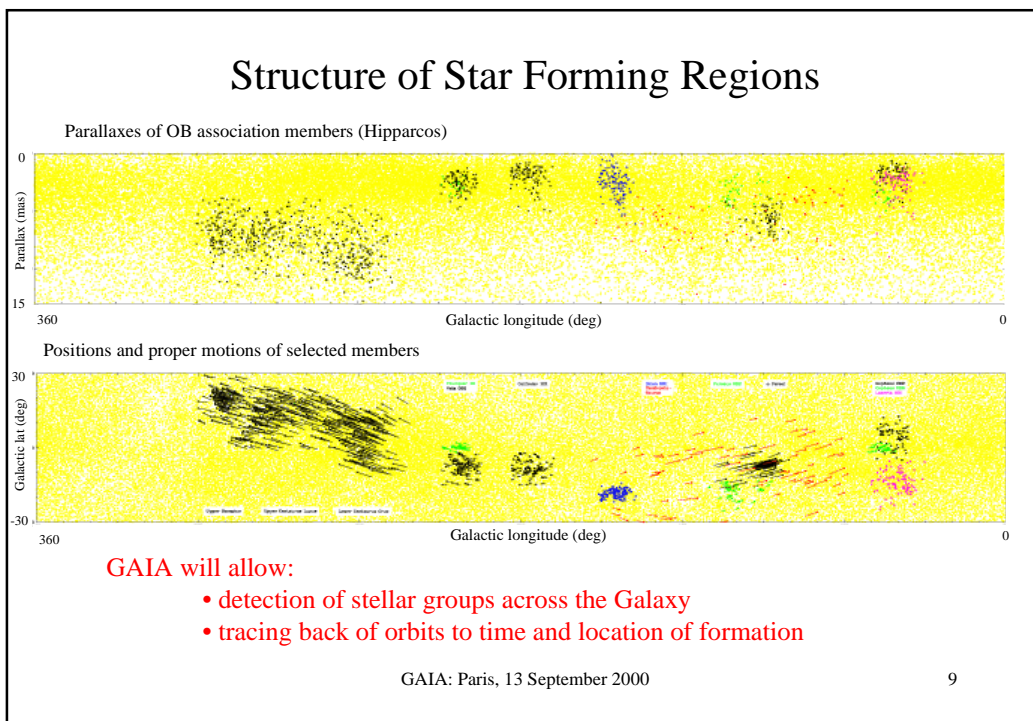
GAIA: Complete, Faint, Accurate

	Hipparcos	GAIA
Magnitude limit	12	20-21 mag
Completeness	7.3 – 9.0	~20 mag
Bright limit	~0	~3-7 mag
Number of objects	120 000	26 million to V = 15 250 million to V = 18 1000 million to V = 20
Effective distance limit	1 kpc	1 Mpc
Quasars	None	$\sim 5 \times 10^5$
Galaxies	None	$10^6 - 10^7$
Accuracy	~1 milliarcsec	4 μ arcsec at V = 10 10 μ arcsec at V = 15 200 μ arcsec at V = 20
Broad band	2-colour (B and V)	4-colour to V = 20
Medium band	None	11-colour to V = 20
Radial velocity	None	1-10 km/s to V = 16-17
Observing programme	Pre-selected	On-board and unbiased

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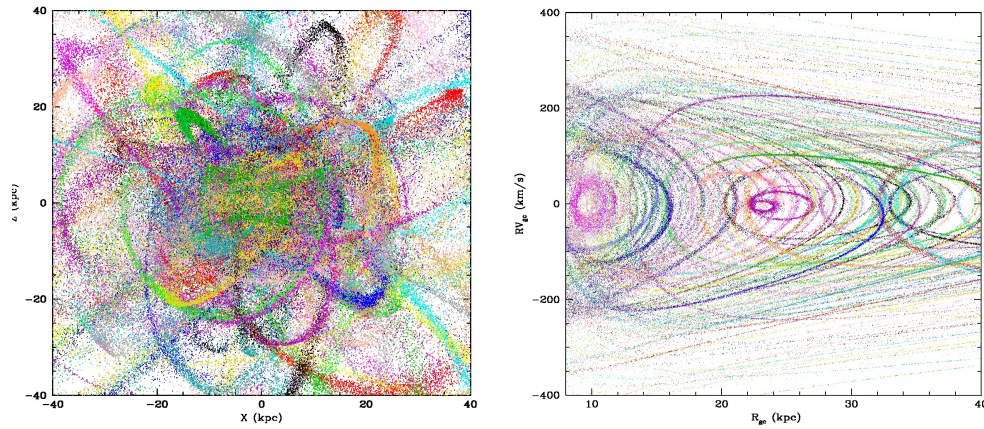
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Tidal Streams in the Galactic Halo

(simulation of accretion of 100 satellite galaxies)

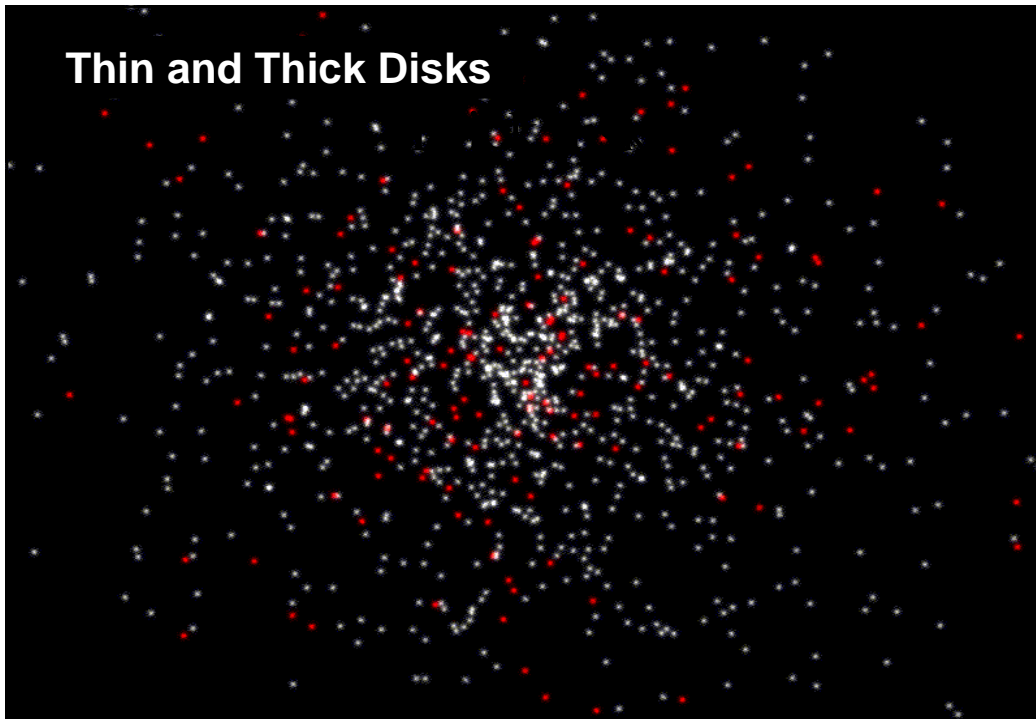


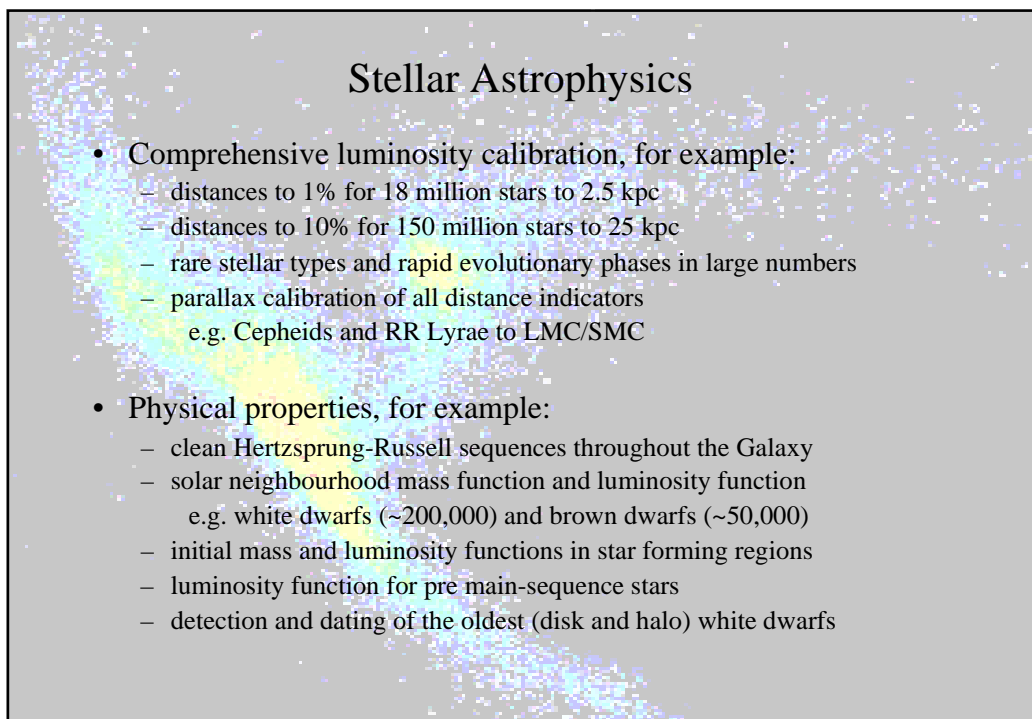
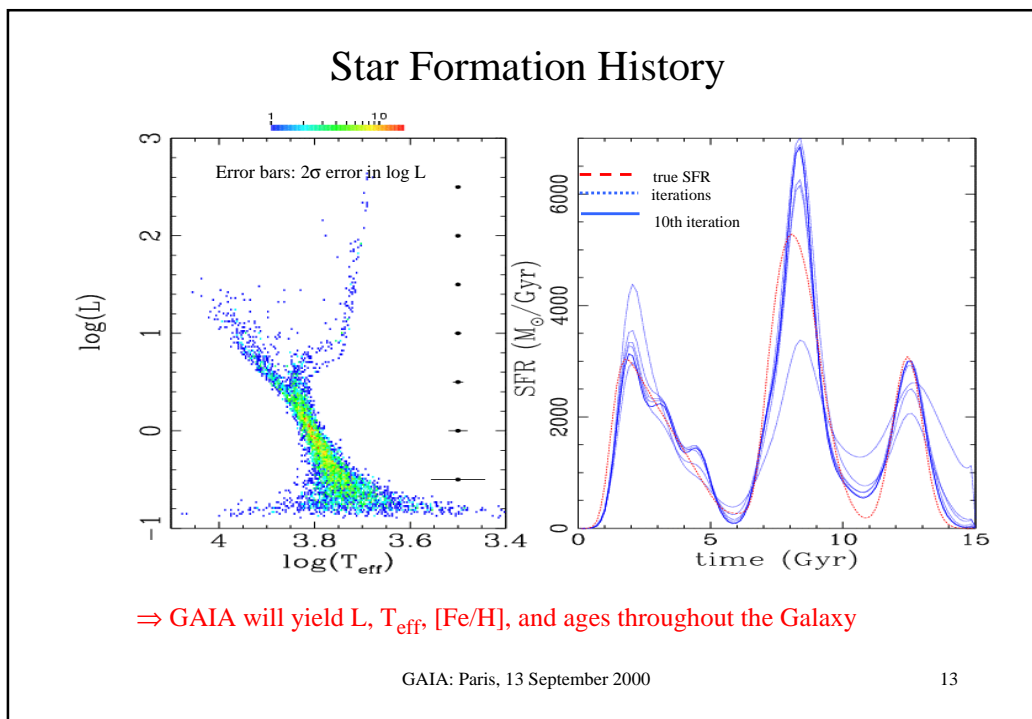
⇒ GAIA will identify details of phase-space substructure

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Thin and Thick Disks





Binary and Multiple Stars

- Constraints on star formation theories
- Orbits for $> 100,000$ resolved binaries (separation > 20 mas)
- Masses to 1% for $> 10,000$ objects throughout HR diagram
- Full range of separations and mass ratios
- Interacting systems, brown dwarf and planetary companions

Photocentric motions:

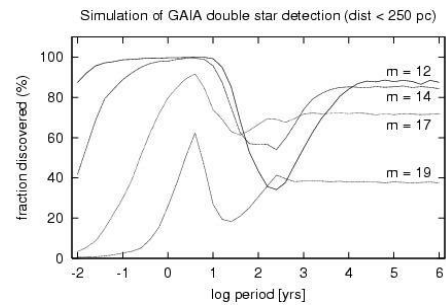
$\sim 10^8$ binaries

Photometry:

$> 10^6$ eclipsing binaries

Radial velocities:

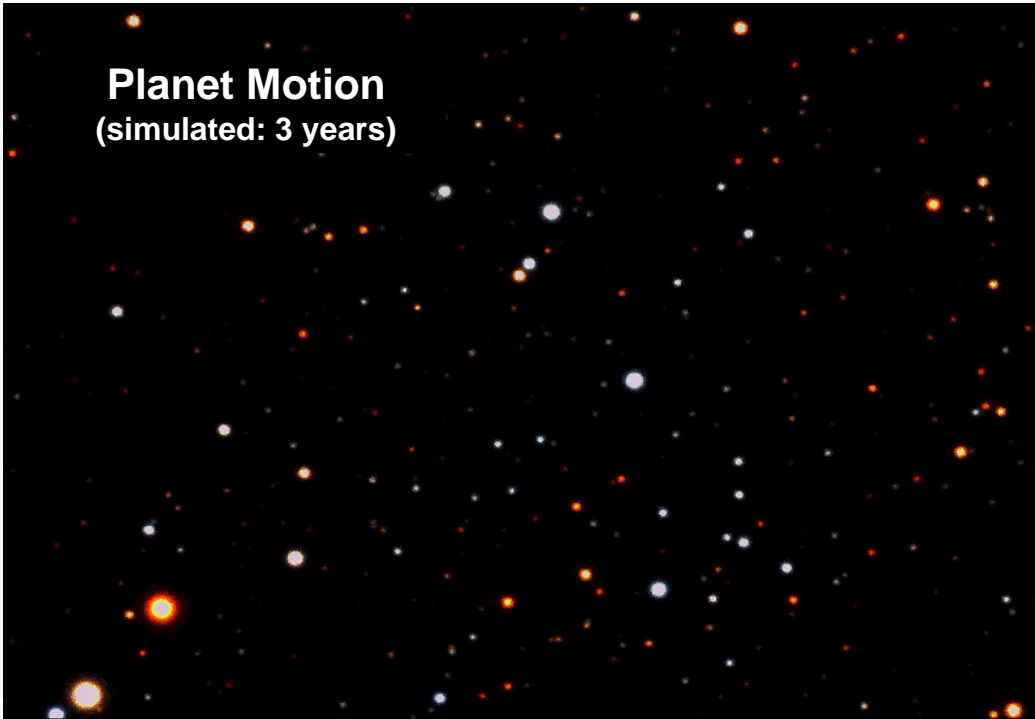
$> 10^6$ spectroscopic binaries



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Planet Motion
(simulated: 3 years)



GAIA: Discoveries of Extra-Solar Planets

- Large-scale detection and physical characterisation
- 20,000- 30,000 giants to 150-200 pc
 - e.g. 47 UMa: astrometric displacement $360 \mu\text{s}$
- complete census of all stellar types ($P = 2\text{-}9$ years)
- masses, rather than lower limits ($m \sin i$)
- orbits for many (≈ 5000) systems
- relative orbital inclinations for multiple systems
- mass down to $10 M_{\text{Earth}}$ to 10 pc

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GAIA: Studies of the Solar System

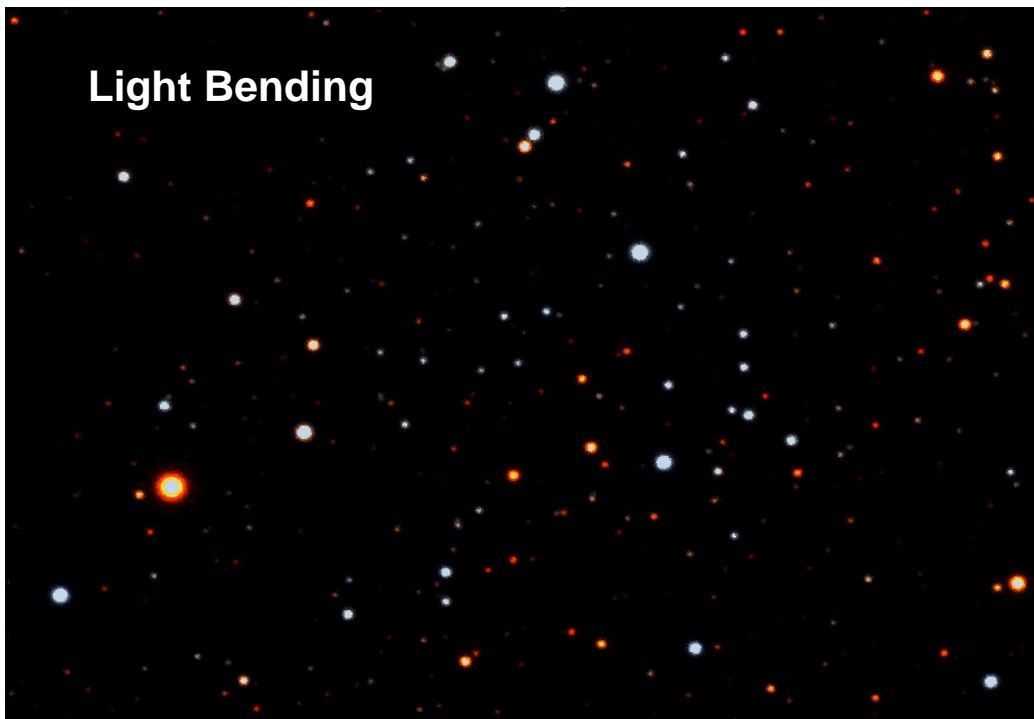
- Deep and uniform detection of all moving objects:
 - ◻ • complete to 20 mag
 - discovery of $\sim 10^5 - 10^6$ new objects (cf. 65,000 presently)
 - taxonomy and mineralogical composition versus heliocentric distance
 - diameters for ~ 1000 asteroids
 - masses for ~ 100 objects
 - orbits: 30 times better than present, even after 100 years
 - Trojan companions of Mars, Earth and Venus
 - Edgeworth-Kuiper Belt objects: ~ 300 to 20 mag + binarity + Plutinos
 - Near-Earth Objects:
 - e.g. Amors, Apollos and Atens (442: 455: 75 known today)
 - ~ 1600 Earth-crossing asteroids > 1 km predicted (100 currently known)
 - GAIA detection: 260 - 590 m at 1 AU, depending on albedo

Galaxies, Quasars, and the Reference Frame

- Parallax distances, orbits, and internal dynamics of nearby galaxies
- Galaxy survey, including large-scale structure
- ~500,000 quasars: kinematic and photometric detection
- ~100,000 supernovae
- Ω_M , Ω_Λ from multiple quasar images (3500 to 21 mag)
- Galactocentric acceleration: $0.2 \text{ nm/s}^2 \Rightarrow \Delta(\text{aberration}) = 4 \text{ } \mu\text{as/yr}$
- Globally accurate reference frame to $\sim 0.4 \text{ } \mu\text{as/yr}$

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General Relativity/Metric

- From positional displacements:
 - γ to 5×10^{-7} (cf. 10^{-5} presently) \Rightarrow scalar-tensor theories
 - effect of Sun: 4 mas at 90° ; Jovian limb: 17 mas; Earth: $\sim 40 \mu\text{as}$
- From perihelion precession of minor planets:
 - β to 3×10^{-4} - 3×10^{-5} ($\times 10$ -100 better than lunar laser ranging)
 - Solar J_2 to 10^{-7} - 10^{-8} (cf. lunar libration and planetary motion)
- From white dwarf cooling curves:
 - dG/dT to 10^{-12} - 10^{-13} per year (cf. PSR 1913+16 and solar structure)
- Gravitational wave energy: $10^{-12} < f < 10^{-9}$ Hz
- Microlensing: photometric (~ 1000) and astrometric (few) events
- Cosmological shear and rotation (cf. VLBI)

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Summary

GAIA will determine:

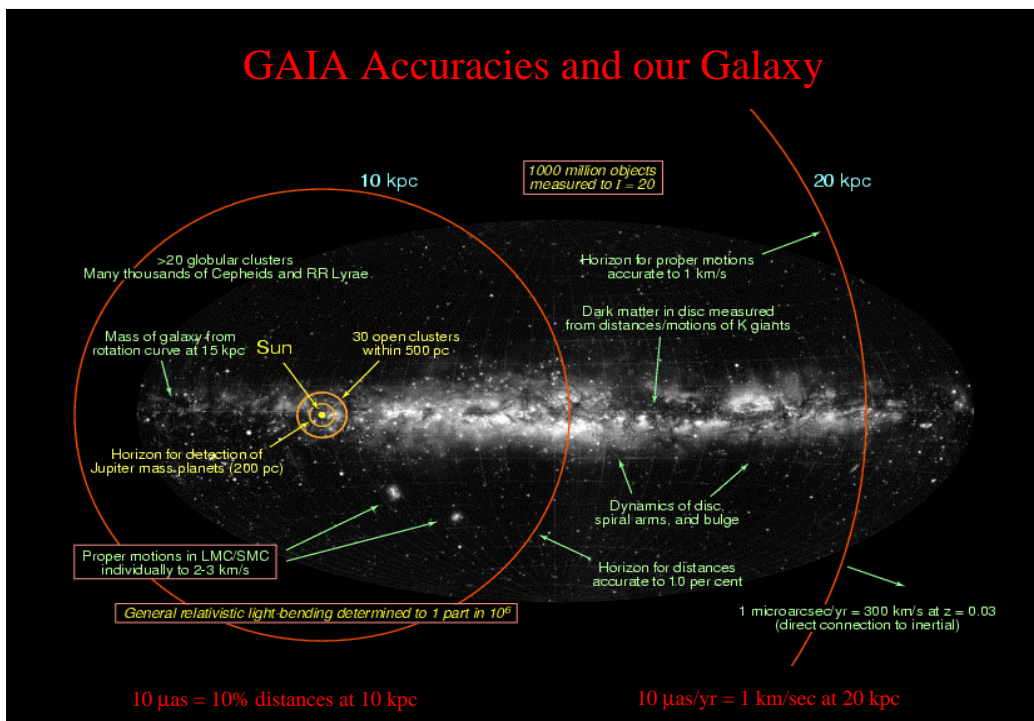
- when the stars in the Milky Way formed
- when and how the Milky Way was assembled
- how dark matter in the Milky Way is distributed

GAIA will also make substantial contributions to:

- stellar astrophysics
- Solar System studies
- extra-solar planetary science
- cosmology
- fundamental physics

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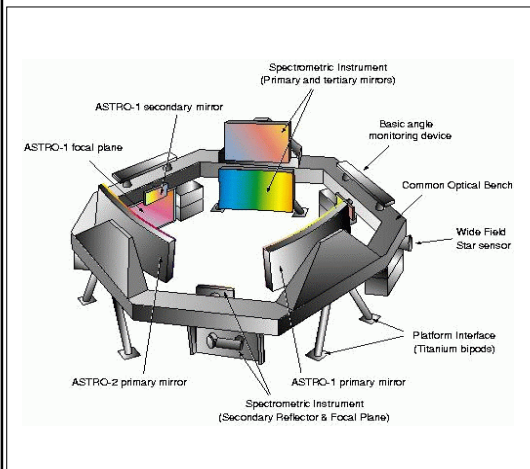
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Design Considerations

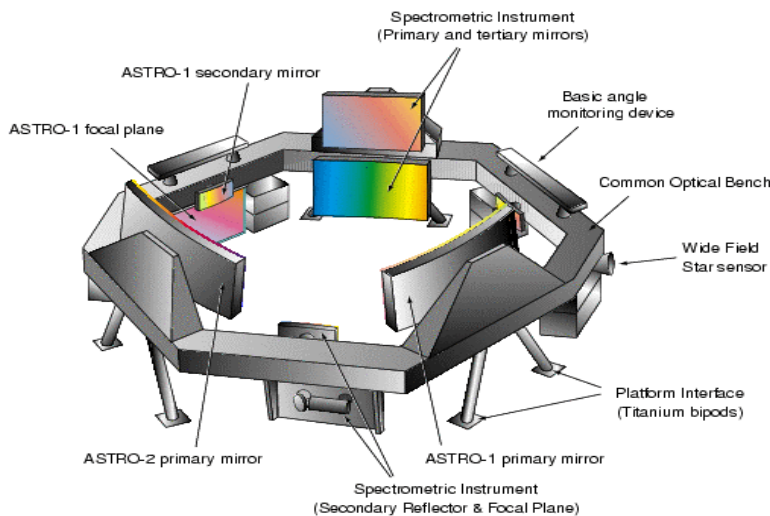
- Astrometry ($V < 20$):
 - completeness \Rightarrow on-board detection
 - accuracies: 10 μas at 15 mag (Survey Committee + science)
 - scanning satellite, two viewing directions
 - \Rightarrow global accuracy, optimal with respect to observing time
 - windowing reduces data rate from 1 Gbps to 1 Mbps
- Radial velocity ($V < 17-18$):
 - third component of space motion
 - measurement of perspective acceleration
 - astrophysical diagnostics, dynamics, population studies
- Photometry ($V < 20$):
 - astrophysical diagnostics (4-band + 11-band) + chromatic correction
 - \Rightarrow extinction; $\Delta T_{\text{eff}} \sim 200 \text{ K}$, $[\text{Fe}/\text{H}]$ to 0.2 dex

Astrophysically Driven Payload



- Two astrometric instruments:
 - field of view = $0.6^\circ \times 0.6^\circ$
 - separation = 106°
- Monolithic mirrors: 1.7 m \times 0.7 m
- Non-deployable, 3-mirror, SiC optics
- Astrometric focal planes: TDI CCDs
- Radial velocity/photometry telescope
- Survey principles:
 - revolving scanning
 - on-board detection
 - complete and unbiased sample

Payload Configuration



Sky Scanning Principle

The diagram illustrates the scanning principle of the GAIA mission. It shows a sphere representing the sky with a central Sun. A spin axis is shown at a 55-degree angle to the Sun. The scan rate is 120 arcsec/s and the spin period is 3 hours. The diagram also shows the scan path as a great circle, with two lines of sight (LOS1 and LOS2) and a star pattern.

Spin axis 55° to Sun
 Scan rate: 120 arcsec/s
 Spin period: 3 hours

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Astrometric Focal Plane

The diagram shows the astrometric focal plane with three main components: Sky mapper, Star pattern detection (TDI mode), and Wide band photometer. The Sky mapper detects the object to be observed. The Star pattern detection (TDI mode) detects the star pattern. The Wide band photometer measures the star's brightness. The diagram also shows the star motion and the scan rate.

pattern #1 pattern #17

Sky mapper (detects the object to be observed) Star pattern detection (TDI mode) Wide band photometer

Sky mapper:

- detects all objects to 20 mag
- rejects cosmic-ray hits
- mag and x,y to main field

Main field:

- area: 0.3 deg²
- size: 60 × 70 cm²
- Number of CCD chips: 136
- CCDs: 2780 x 2150 pixels

Pixels:

- size: 9 x 27 μm²
- window area: 6 x 8 pixels
- flush frequency: 15 MHz
- readout frequency: 30 kHz
- total read noise: 6e⁻ rms

Broad-band photometry:

- 4 colour

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On-Board Object Detection

Requirements:

- unbiased sky sampling (mag, colour, resolution, etc)
- no all-sky catalogue at GAIA resolution (0.1 arcsec) to $V \sim 20$

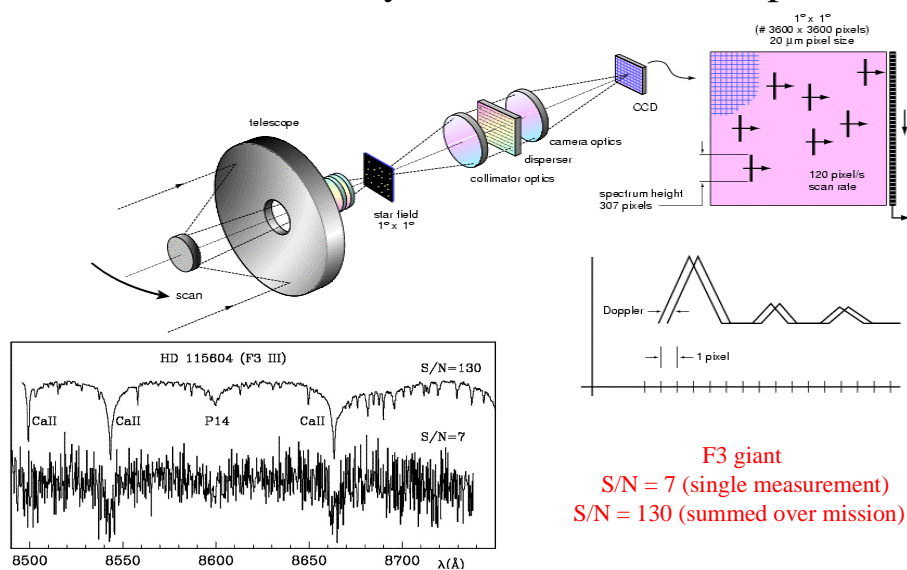
Solution: on-board detection:

- no input catalogue or observing programme
- good detection efficiency to $V \sim 21$ mag
- low false detection rate, even at very high star densities
- maximum star density: ~ 3 million stars/deg² (Baade's Window)

Will therefore detect:

- variable stars (eclipsing binaries, Cepheids, etc)
- supernovae: 10^5 expected
- microlensing events: ~ 1000 photometric; ~ 100 astrometric
- Solar System objects, including near-Earth asteroids and KBOs

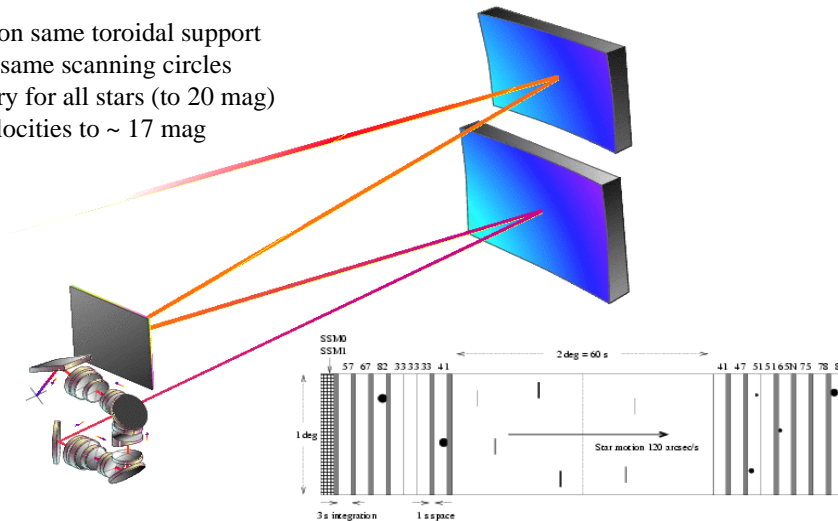
Radial Velocity Measurement Concept



F3 giant
S/N = 7 (single measurement)
S/N = 130 (summed over mission)

Radial Velocity and Photometric Instrument

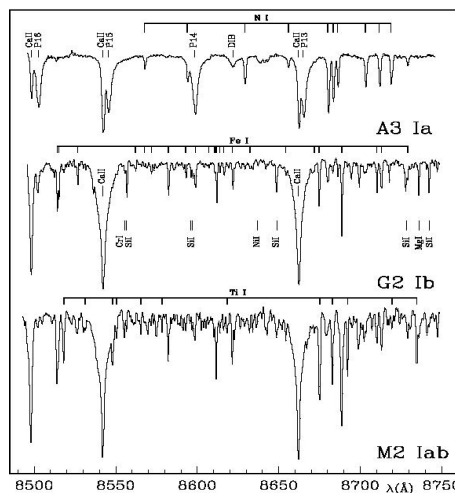
- Mounted on same toroidal support
- Observes same scanning circles
- Photometry for all stars (to 20 mag)
- Radial velocities to ~ 17 mag



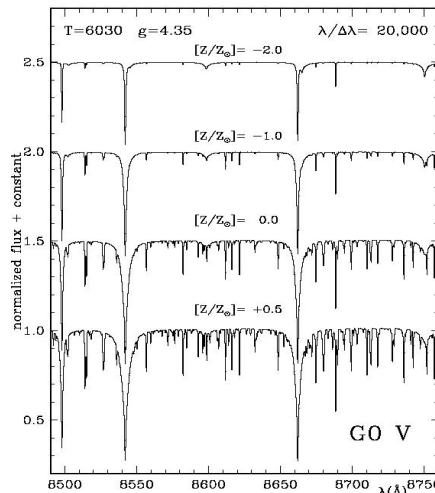
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Spectral Sequences around Ca II



Effect of temperature: A to M stars



Effect of metal abundance in G stars

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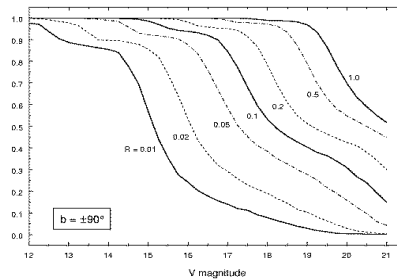
Astrometric Accuracy

G (~V mag)	10	11	12	13	14	15	16	17	18	19	20	21
Parallax	4	4	4	5	7	11	17	27	45	80	160	500
Position	3	3	3	4	6	9	15	23	39	70	140	440
Annual proper motion	3	3	3	4	5	8	13	20	34	60	120	380

5-year accuracies in μas

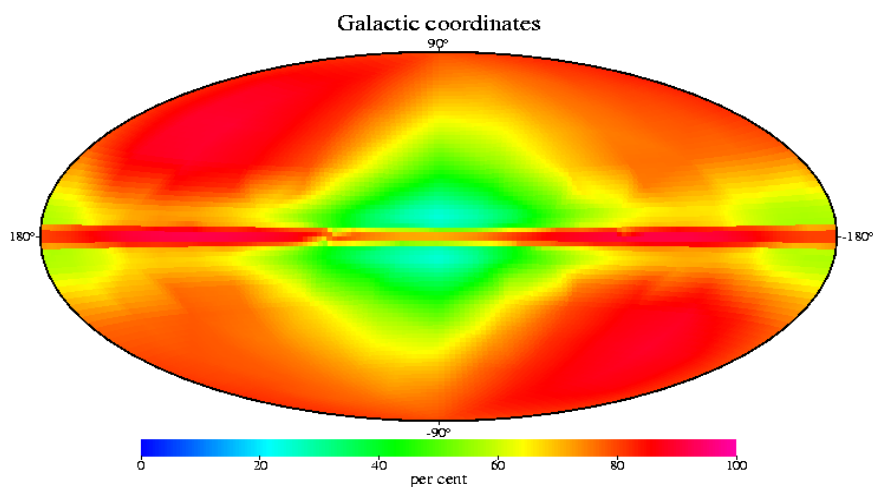
Derived from comprehensive analysis:

- image formation (polychromatic PSF)
- evaluation versus spectral type/reddening
- comprehensive detector signal model
- sky background and image saturation
- attitude rate errors and sky scanning
- on-board detection probability
- on-ground location estimation
- error margin of 20 per cent included
- results folded with Galaxy model

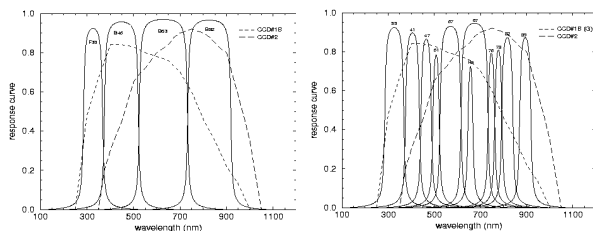


Fraction with given relative parallax error towards Galactic poles

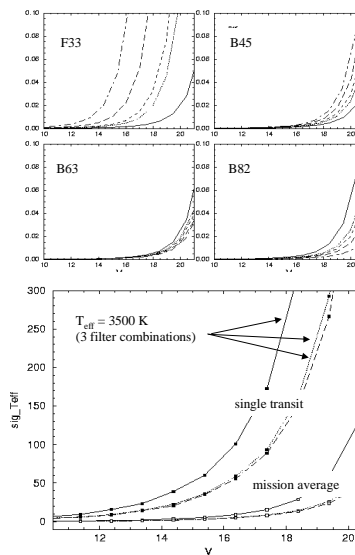
Accuracy Example: Stars at 15 mag with $\sigma_{\pi}/\pi \leq 0.02$



Photometric Accuracies and Diagnostics



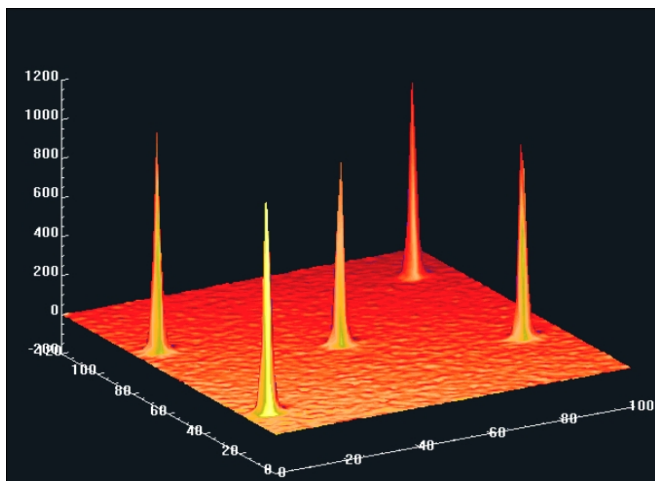
- **Photometric system (above):**
 - optimised for astrophysical diagnostics
- **Photometric accuracy assessment (top right):**
 - photon noise, sampling, CCD response, etc.
 - single transit + mission average (100 transits)
- **Astrophysical diagnostics (right):**
 - ⇒ reddening, T_{eff} , [Ti/H], [M/H], etc.



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CCD Centroiding Test Results



Astrium contract (Sep 2000)
 'GAIA-mode' operation
 EEV CCD 42-10
 3-phase, 13 μ m pixels
 Illumination: 240,000 e⁻

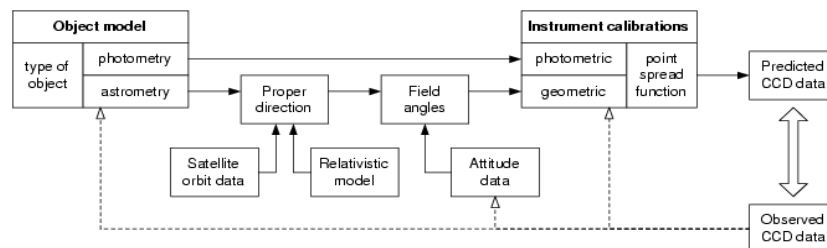
Frequencies:
 TDI: 2.43 kHz
 Dump: 4 MHz
 Binning: 1 MHz
 Readout: 90 kHz

Differential centroid errors:
 rms = 0.0038 pixels
 ($\approx 1.2 \times$ theoretical optimum)

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Data Analysis: Concept and Requirements



Capacity: ~100 Terabytes

Overall system: centralised global iterative approach

Accessibility: quasi-random, in temporal and object domains

Processing requirements: entire task is $\sim 10^{19}$ flop

Numerical: 0.1 microarcsec = 10^{-13} of a circle (64-bit marginal)

Data base structure: e.g. Objectivity (cf. Sloan)

Results: time-critical results available early (NEO, supernovae etc)

⇒ **Prototype:** Hipparcos global astrometry re-reduced during concept study

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Main Performances and Capabilities

Accuracies:

- 4 μ as at $V = 10$ 10 μ as at $V = 15$ 0.2 mas at $V = 20$
- radial velocities to few km/s complete to $V = 17.5$
- sky survey at ~ 0.1 arcsec spatial resolution to $V = 20$
- multi-colour multi-epoch photometry to $V = 20$
- dense quasar link to inertial reference frame

Capabilities:

- 10 μ as \equiv 10% at 10 kpc \equiv 1 AU at 100 kpc
- 10 μ as/yr at 20 kpc \equiv 1 km/s
- ⇒ every star in the Galaxy and Local Group will be seen to move
- ⇒ GAIA will quantify 6-D phase space for over 300 million stars, and 5-D phase-space for over 10^9 stars

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Main Payload Requirements (1/2)

Orbit:	Earth/Sun straylight	minimised
	Earth/Moon occultation	minimised
	Thermal/radiation impact	minimised
	Eclipse during observation	avoided
Thermal:	Optical bench stability	few tens of μK
	CCD temperature	$\approx 200\text{ K}$
Mechanical:	Mechanical and dynamic interference minimised	
Outage:	Mission outages minimised	
Lifetime:	5 years nominal; 6 years extended	

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Main Payload Requirements (2/2)

- Payload composed of:
 - two identical astrometric telescopes:
 - separated by 106°
 - knowledge to $1\ \mu\text{as}$ over one revolution (3 h)
 - spectrometric telescope:
 - medium-band photometer
 - radial-velocity spectrometer
- Astrometric accuracy:
 - $< 10\ \mu\text{as}$ rms for $V = 15\ \text{mag}$
 - complete between $V = 3\text{--}20\ \text{mag}$
- Stars measured in Time-Delayed-Integration (TDI) over 17 CCDs
- Star profiles, along scan, generated at 1 Mbps

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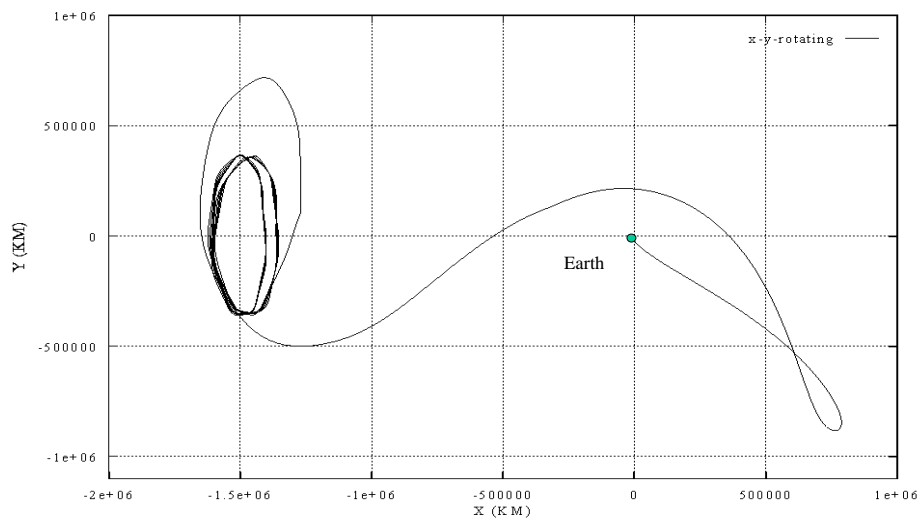
Launch and Operational Orbit Strategy

- Operational orbit - around Lagrangian L2 of Sun-Earth system
- Transfer orbit duration -220 to 260 days, according to launch date
- Launcher - Ariane 5 in dual/multiple configuration
- GAIA location - within SPELTRA, as lower passenger
- Launch strategy - Ariane 5 injection into a standard GTO orbit
- transfer orbit and final injection around L2 by autonomous satellite propulsion system
(option: to use Ariane 5 third stage)
- Launch window - daily window compatible with Ariane 5
midnight window for dual launches
- Launch date - 2009

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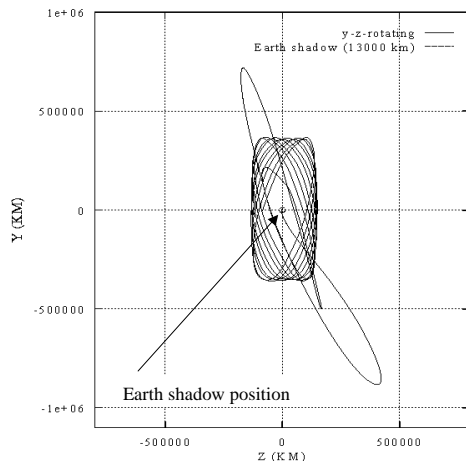
Transfer Orbit



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Operational Orbit

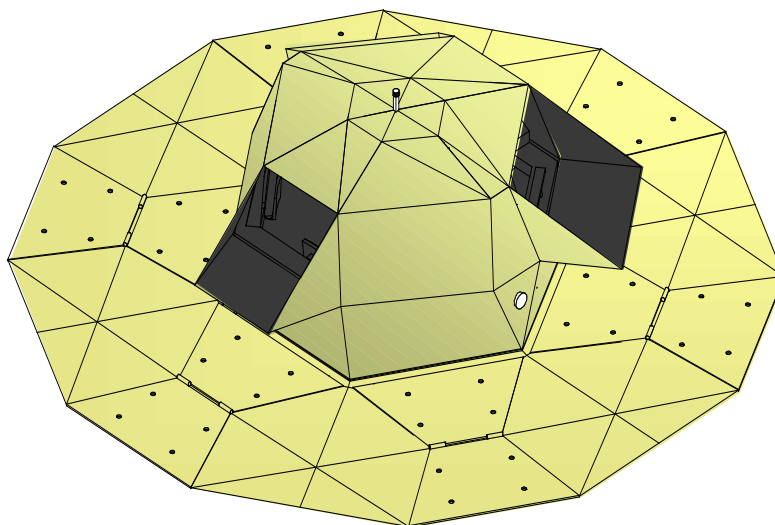


- L2, Lissajous
- Semi-axes:
400.10⁶ km × 100.10⁶ km
- Orbit period: 6 months
- Sun-Satellite-Earth angle:
≤ 15° (40° to 70°)
- With one avoidance manoeuvre,
eclipse-free condition kept for
much longer than 5 years (~12y)
- Ground station visibility:
8 hrs/day (Perth)

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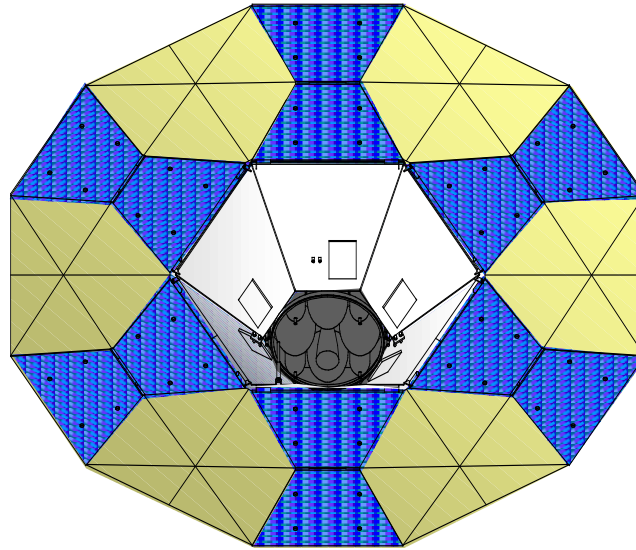
Spacecraft: Top View



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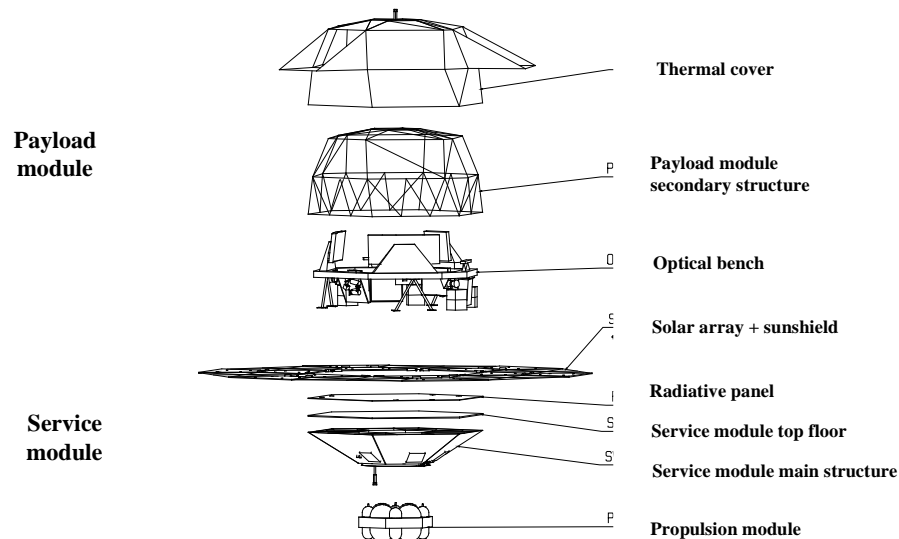
Spacecraft: Bottom View



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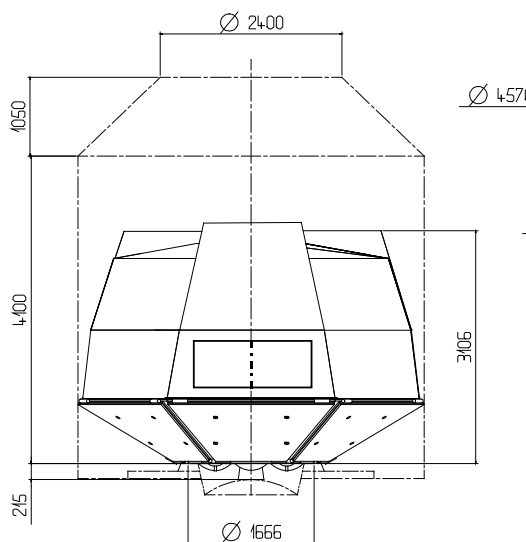
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Satellite Exploded View



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Spacecraft: Undeployed Configuration



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Spacecraft Design Approach (1/5)

- Mission Lifetime 5 years nominal, >4 years observation
- Solar Aspect Angle >120° (payload protection from Sun)
- Spacecraft stabilised 3-axis, with 120 arcsec/sec scanning law (1 revolution every 3 hr)
- Lift-off Mass 3137 kg, with autonomous propulsion
 (with 20% system margin) 2337 kg, without propulsion
- Power 2468 W at 5-year end-of-life
 (with 10% system margin) 2616 W at 6-year end-of-life
- Pointing Accuracy (3 σ):
 - absolute pointing error < 5 arcmin
 - relative pointing error < 0.002 arcsec/sec (1 σ)

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Spacecraft Design Approach (2/5)

- Configuration Modular: service module (SVM) and payload module (PLM) thermo-mechanically decoupled
- SVM structure Truncated hexagonal pyramid shape, to avoid turning shadow on sunshield, with six lateral walls to support SVM (and PLM) electronics
- PLM structure Monolithic, toroidal optical bench of SiC, with 3+3 SiC mirrors and three focal plane assemblies. 3 isostatic connections with SVM
- Stabilisation 3-axis attitude control with star sensor, coarse sun sensor, 1-axis gyro and 6 redundant, 10N bi-propellant thrusters
- Attitude control Continuous scanning by 1 mN FEEP thrusters

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Spacecraft Design Approach (3/5)

- Propulsion Bi-propellant system, with single 400 N engine, integrated by the 6x10 N thrusters for orbit correction, final orbit injection and maintenance
Four standard propellant tanks; two pressurant
- Thermal Control Passive, with heaters, ensuring efficient payload stability and PLM/SVM thermal decoupling
- Power Supply Solar array: deployable, six wings of 2 GaAs panels each (24 m²), within annular sunshield of 4.5 m inner and 8.5 m outer diameters
Regulated power bus: 28 V, with two Li-ion 14Ah batteries for eclipses during launch and transfer phases (no eclipses in operational orbit)

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Spacecraft Design Approach (4/5)

- Data Handling
 - On-Board Data Handling, with packet Telemetry and Telecommand
 - Centralised control unit: spacecraft operation, attitude and orbit control, and thermal control
 - 100 Gb mass memory, allowing for science data rate dumping at 3 Mbps (1 Mbps continuous payload data rate)
- Communications Standard ESA X-band up- and down-links; 2 kbps for housekeeping and omni-directional coverage (2 low-gain antennas and 17W-RF)
- Science telemetry: X-band down-link at 3 Mbps (typical); six electronically-scanning phased-array antennae (EIRP > 32 dBW)

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Spacecraft Design Approach (5/5)

Budgets	Mass (kg)	Power (W)
Payload module	893	1527
Service module	895	717
Margin	339	224
Propulsion	1010	
Total	3137	2468

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Identified Key Technology Activities

- Validation of CCD performance and CCD development
- Focal plane assembly, detection and data handling electronics
- Large size silicon-carbide mirrors ($1.7 \times 0.7 \text{ m}^2$)
- Ultra-stable large size SiC structures for payload optical bench
- Large deployable solar array/sunshield assembly
- High-stability optical benches (basic angle verification)
- Phased-array antenna for high data rates and far orbits
- Optimised on-board compression algorithm
- Ground calibration/verification approach and facilities
- Database architecture

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Cost at Completion

(current Cornerstone 5 Budget Envelope = 541.7 MEuro)

Project Cost Estimate	MEuro (EC 2000)
Procurement Cost (ESA + Industry + Overheads + Contingency)	413.8
Spacecraft Operations	35.3
Science Operations	12.9
Launch	111.9
Total Project Cost Estimate	573.9

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